

# Firefleye

## Tunnel Modeling Through Throwable Illumination

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## Abstract

Tunnels and caves on the moon surface are completely dark. This makes autonomous navigation a challenge. One solution for this problem is to carry a light source, on the robot. However, this is not a good solution as light intensity drops as  $1/r^4$  for area light sources.

This paper introduces a concept of illuminating dark tunnels via throwable lights known as Firefleyes. Unlike LIDARs, which provide only geometrical information via 3D point clouds, the method discussed in this paper provides both geometrical and texture information of dark tunnels.

## 1 Introduction

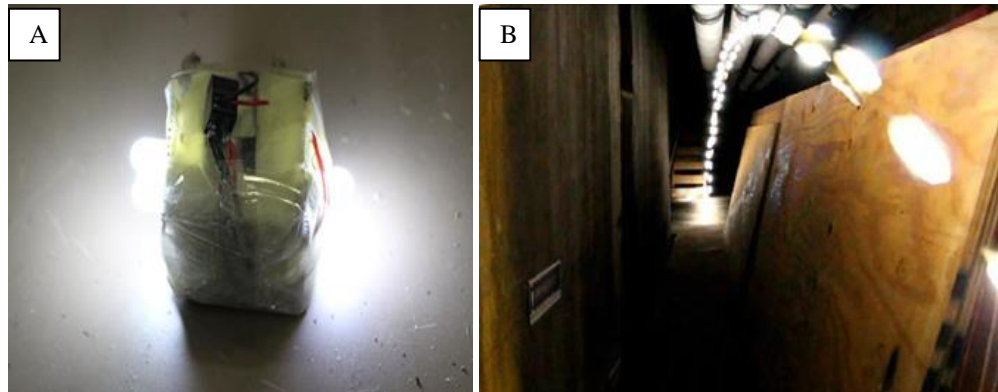
33 The surface of the moon has frequently been explored by rovers and orbiting spacecraft, but  
34 the subterranean features have remained unexplored. Recent discoveries of lava tubes and  
35 skylights on the moon with access to the lava tubes have brought to light the possibility of sending  
36 a mission to explore below the lunar surface. These underground lunar caves are thought to  
37 contain ice, and could serve as havens from meteorites, radiation, and thermal extremes.

38 Underground missions would be devoid of natural light. To navigate adequately in a dark  
39 cave, an imaging system will have to be developed to better illuminate the area. The purpose of  
40 the Firefleye project is to evaluate the concept of a light-throwing system that could generate a  
41 well-lit image of a dark cave under the lunar surface.

42 The Firefleye system involves a launcher that propels a bright light down a tunnel of unknown  
43 size and features. A camera would record a video of the cave as the projectile is launched into it.  
44 Software would then composite all of the recorded frames to combine the brightly illuminated  
45 areas of each into one well-lit image of the cave and a depth map of the area. This data on what  
46 lies ahead of the future rover could potentially be more useful and efficient than the limited depth  
47 only values that LIDAR has to offer. A comprehensive series of tests have been performed using  
48 several trajectories and light sources to determine the optimal components to be used in any future  
49 implementations of Firefleye.

## 50 2 Prior Work

51 The original work that inspired this project was carried out by Uland Wong. In his  
52 preliminary test, Uland made a crude light source, shown in Figure 1A, and recorded several  
53 images as the light source traveled in a tunnel. He then took the recorded images and applied  
54 image fusion to get a result shown in Figure 1B.



55  
56 Figure 1: (A) Light Source from Uland's Experiment. (B) Image fusion result from Uland's Experiment

57 Though this novel idea produced good preliminary results, it has two main issues. First, this  
58 experiment uses a crude non-diffused and non-symmetrical light source. This implies that the light  
59 source wouldn't behave like a point source, which makes image post-processing difficult. In  
60 addition, the light source was only bright enough to light a 0.5 m by 6m by 2 m tunnel (after  
61 image fusion).

62 Second, the right side wall of the tunnel was made up of a non-Lambertian material. In other  
63 words, the reflection from that side of the wall is not angle invariant; therefore, the resulting image  
64 varies with the camera's view point.

65

66

## 67 3 Methods

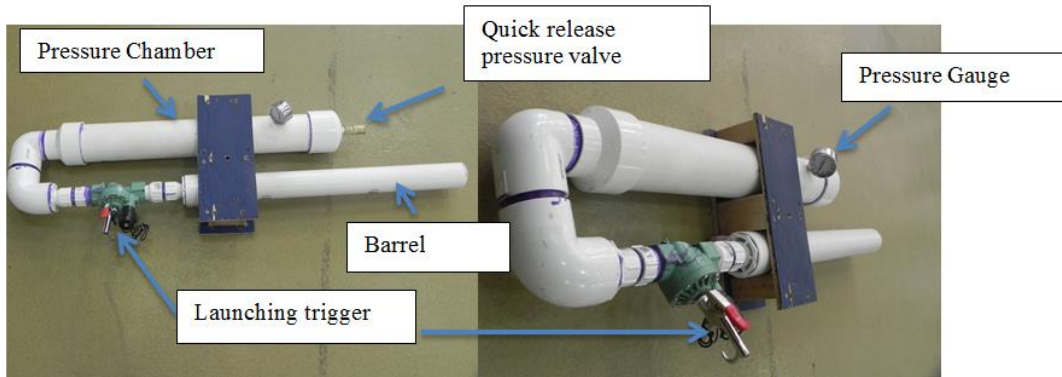
68

69 Several stages of setup and preprocessing were done before the analysis of projectiles in a  
70 video was done. Description of each step is given in the following sections.

71 **3.1 Projectile Launcher**

72 The first step of the methodology followed in this experiment was to properly setup the  
73 projectile launcher. The projectile launcher used was a spud gun with a 2 ft long and 3 in internal  
74 diameter pressure chamber, 2 ft long and 2 in internal diameter barrel, and a modified sprinkler  
75 valve. Traditional spud guns use a ball valve as a release mechanism for the compressed air in the  
76 pressure chamber. Most ball valves are not designed to be rotated easily which makes automating  
77 the projectile launcher mechanism a challenge. However, with the modified sprinkler valve  
78 system, the projectile launcher mechanism can be shot with less than 1N force.  
79

80 To achieve a successful projectile, the correct pressure and the angle of launch had to be set.  
81 These values could change based on the desired range of the projectile. The default setup for our  
82 experiment was 10 degrees launching angle and 15 PSI of pressure, which gave us a projectile  
83 with a maximum height of 6 ft and a range of 60 ft.



84  
85 Figure 2: Image of Projectile Launcher (the term “cannon” is also used interchangeably throughout this paper)

86 **3.2 Camera setup**

87 To record the full flight of the projectile a video camera was set next to the projectile launcher on  
88 a tripod. To get maximum number of photons in dark environment, the aperture size was set to  
89 maximum and all automatic features were switched to manual mode.  
90



91  
92  
93 Figure 3: Testing assembly consisting of cannon mount, compressor, and cameras.

94 **3.3 Image analysis**

95 Generating fully lit image and depth map requires several stages of processing. The first  
96 step performed in this project was the pre-processing stage that included converting videos to  
97 image frames and converting RGB images to grayscale. Following that, Image fusion and post  
98 processing of light source extraction were carried out finally leading to a good estimation of a  
99 depth map. The following subsections go through each process in brief detail.

100 **3.3.1 Video to frame conversion**

101 To analysis each frame separately; a MATLAB based algorithm was used to convert the videos  
102 recorded into separated images where each image corresponds to one frame in the video.

103 **3.3.2 Image Fusion**

104 Image Fusion is a process of combining several images into one. This process is done in a way  
105 that the fused image would be more informative than each individual images used to create it. In  
106 this paper, this was achieved via taking the maximum pixel over all the frames.

107 **3.3.3 Light source extraction**

108 The final result includes a fully lit tunnel. However, the image fusion method fails to remove the  
109 light source from the fused image; therefore, post-processing needs to be done to remove the light  
110 source from the fused image. This was achieved by tracking the position of the light source and  
111 replacing the pixel values with the mean pixel value of the fused image.

112 **3.3.4 Depth Map**

113 Depth map is an image that represents distances of different objects in a scene form a specified  
114 view point. For the Firefleye project computing the depth map was possible due to the principle  
115 related to Lambertian reflectance (see Appendix 8.1.6).  
116 The input image was a three dimensional matrix where the rows and columns correspond to pixels  
117 in one frame and the depth corresponds to the number of frames. Depth map for a given run was  
118 found by recording the indices corresponding to maximum pixel values. Since each frame  
119 correspond to different time steps in the projectiles flight, the frame number could be taken as the  
120 depth of the projectile (the distance from the camera to where it is in the tunnel).

121

122 **4 Results**

123

124 The performance of system considered in this paper, which includes projectile  
125 launcher, and image analysis algorithms was tested and analyzed separately.

126

127 **4.1 Projectile Launcher Results**

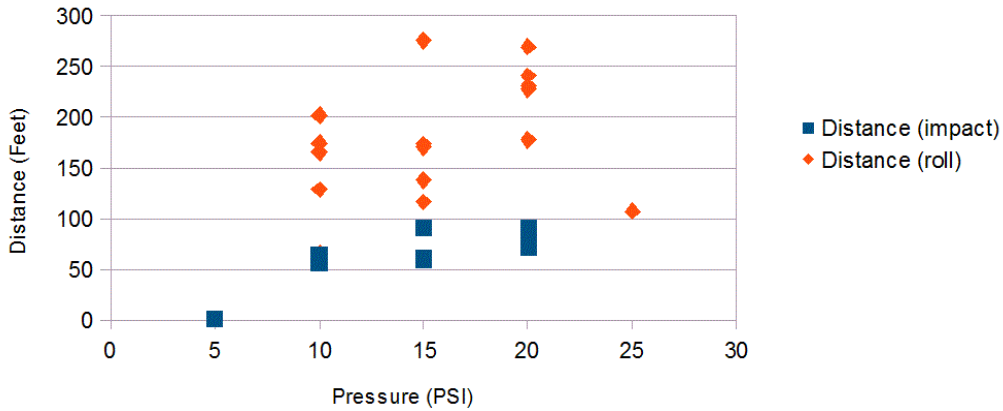
128

129 The following results came from a field test in a 300ft+ mining tunnel that  
130 was less than 6 feet tall. As displayed above.

131

132 From these tests, the following results were recorded:

133



134 Figure 4: Pressure vs. distance traveled by the projectile at 10 degrees launching angle. The impact distance is the distance  
 135 the projectile flew in a 10 foot wide by 6 feet tall mining tunnel. This distance is upon first impact of the ground. The  
 136 rolling distance is the distance the projectile traveled until rest. There are far less impact distance measurements as once the  
 137 lights were turned off in the mining tunnel, it is very difficult to judge accurate landing position.  
 138  
 139

141 **4.3 Design Matrix and Design Constraints**

142  
 143 Our launcher was chosen using the design matrix given in Table 1. Our launcher  
 144 was chosen for testing on Earth as opposed to withstanding moon-like conditions.  
 145 The description of each criterion is given below.

146 Weight - the overall system weight of implementing a design.

147 Variable speed - possible firing speeds for the projectile launcher.

148 Cost - the relative system cost of implementing a design.

149 Distance - possible firing distance which is important for long tunnels.

150 Space relevance – future applications in space.

151 Ease of automation - the difficulty it would be to fully automate this system.  
 152

153 Table 1: Design Matrix for Projectile Launcher

	Flywheel	Spring	Coil Gun	Crossbow	Pressure Gun
Weight	1	10	1	8	7
Variable speed	9	8	6	8	10
Cost	5	4	2	8	7
Distance (x2)	9	9	2	9	10
Space Relevance (x2)	1	8	4	7	6
Ease of Automation	10	8	2	5	9
Total	5.63	7.35	3.88	7.63	8.13

154  
 155  
 156  
 157 **4.4 Software Results**

158  
 159 Several tests were carried out at the steam tunnels in the basement of Newel Simon Hall  
 160 at Carnegie Mellon University and at Bruceton Coal Mines. The following figures show image  
 161 fusion results for these different tests.



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163  
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Figure 5: Image Fusion result with a sphere light made up of 18 (1 watt) LEDs (left) and Static long exposure image with 10 W head light (right) at at Bruceton Coal Mines



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Figure 6: Image fusion result (12 ft by 6 ft by 60 ft) tunnel with 18, 1 watt LEDs arranged in a plane



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Figure 7: Image fusion result (4 ft by 6 ft by 150 ft) tunnel with 18, 1 watt LEDs arranged in a plane

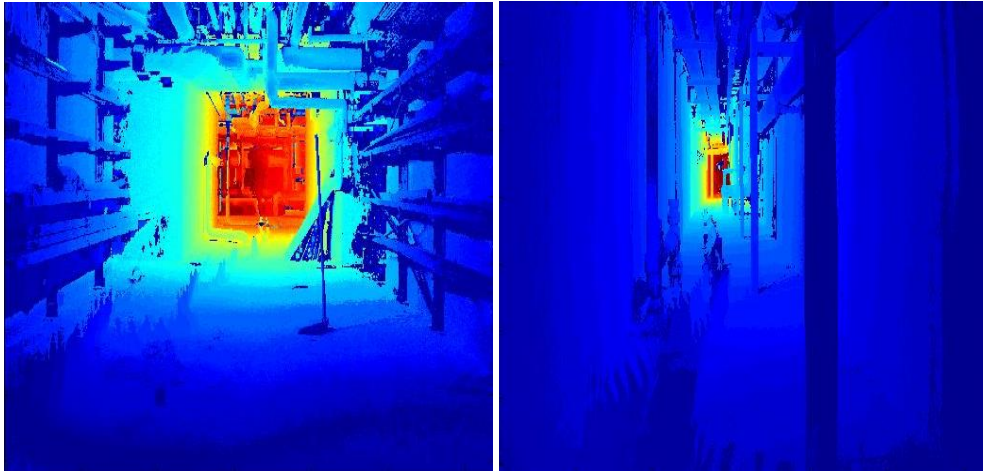


Figure 8: Depth map for Figure 6 (left) and Figure 7 (Right)

172  
173  
174

## 175 **5 Analysis**

### 176 **5.1 Choice of projectile Launcher design**

177 There were two main problems with the projectile. The design of the projectile  
178 was not done well and the LED's draw too much power for any batteries tested. In the  
179 tests carried out, a spherical light source was chosen as the ideal light source required by  
180 assumptions made for the software. In field tests that worked well, a high discharge 12  
181 volt Ni-mh battery with 18- 3 watt LED's was able to light a 4 ft by 6 ft by 60 ft tunnel  
182 (after image fusion) at a current draw of 8.5 amps. These LED's output 100 lumens each.  
183 A small launchable battery that can perform at such a demanding current draw is required  
184 to make launchable lights sufficiently bright to illuminate large areas.

### 185 **5.2 Launcher Analysis (design matrix analysis)**

186 Looking at the design matrix, there were many reasons other design were not  
187 chosen. For the flywheel design, the weight was a big factor as large flywheels were  
188 required and kinetic wheel energy was transferred into the projectile. The spring and  
189 crossbow designs were not chosen for testing due to inefficiency of launching distances  
190 compared to a pneumatic system. The coil gun had a better projectile launching distance  
191 but was rejected because it required about 500 volts to achieve 50 ft of launching  
192 distance. It also failed to pass the space relevance requirement because it releases a lot of  
193 heat and dispensing that would be a challenge in a vacuum environment. Therefore,  
194 considering all the requirements, the pneumatic pressure gun was chosen as the final  
195 results.

196 Looking at the graph of distances fired vs psi, two main points could be inferred.  
197 The first is that the cannon was consistent within 10 feet at the ideal launch angle of 10  
198 degrees and a pressure of 15 psi. This spread can be attributed to a number of small  
199 flaws. The first is a non-air tight seal behind the projectile in the barrel. This leads to  
200 inconsistent power. The second is that the barrel induced spin on a spherical projectile.  
201 This caused a Magnus effect [7] which caused the projectile to move out of its flight path.  
202 A third cause was valve opening inconsistencies. Opening the valve at different speeds  
203 caused the pressure chamber to dump air into the barrel at different flow rates, which in  
204 turn affected the projectile initial velocity and distance traveled. The final cause of

205 inconsistencies was the pressure gauge on our chamber, which was not accurate for small  
206 changes in pressure.

207

### 208 **5.3 Image Fusion and Depth map result analysis**

209

210 Figure 5 through Figure 8 show the outputs for the software algorithm that  
211 computes image fusion and depth map. Figure 5 shows a side by side comparison  
212 of an image fusion with the diffused light source and a long exposure image with  
213 a static head light set behind the camera. As the head light was a focused light,  
214 portions of the image in the long exposure image appear well lit than the image  
215 fusion case. However, looking at each image as a whole, more texture could be  
216 extracted from Figure 5A.

217

218 Figure 6 and Figure 7 are image fusion results from a test using a light  
219 source arranged in a plane. The video was captured with a subject running with  
220 the light source in front of him/her. Even though the image of the subject appears  
221 in each frame, when performing image fusion, only the maximum pixel is taken;  
222 therefore, the image of the subject is automatically removed without a need for  
223 extra post processing.

224

225 Figure 8 shows a depth map where blue indicates closer objects and red  
226 indicates further objects. Comparing Figure 8 with Figure 6 and 7, it is clear that  
227 the depth map was able to capture object details and their relative distances.

228

## 229 **6 Future Work**

230

231 Future work for the Firefleye system will require work on the launcher,  
232 projectile, and software in order to achieve better results in illuminating dark  
233 rooms and tunnels.

234 The launcher discussed in this paper was a testing rig for the throwable  
235 lights. A future launcher will require many changes. To increase consistency, the  
236 launcher will need to be fully automated. An automated system would involve a  
237 pan/tilt mechanism for the launcher and cameras. A mechanism to reload the  
238 cannon, open and close the main valve, and fill the pressure chamber will need to  
239 be developed. By automating all of these tasks with a robust system, the  
240 repeatability and consistency of the cannon will increase which allows more  
241 accurate testing of the cannon. Another main topic that will need to be looked into  
242 is the development of a space rated cannon. A space rated cannon will take many  
243 different considerations that the current cannon did not take which would allow it  
244 to function in a zero atmosphere and near absolute zero temperature  
245 range.

246 The projectiles will also require a number of changes. Although the light  
247 sources were able to withstand being fired from a cannon and repeatably work,  
248 the lights were not bright enough to illuminate large areas. To fix this, a new  
249 electrical system will need to be implemented into the  
250 lights. The main problem of the old system is the lack of a small, high discharge  
251 battery that can supply the necessary energy needed for high power LEDs. Large



252 battery tests of the current LEDs showed that the LEDs themselves illuminated  
253 large tunnels, but scaling down the battery to a compact size caused problems in  
254 the power supply to the LEDs. A new battery/capacitor system will need to be  
255 found or another method of illumination in order to make these projectiles work.  
256 The design of the light itself should also be changed into a light that can be put  
257 together neatly as this will allow the weight to be minimized.

258 The software for this system will need a better implemented depth map. The  
259 current depth map estimation gives relative values of depth in the current depth  
260 map being processed. Future depth map estimation will quantify depth into a real  
261 value that can then be used for modeling dark  
262 spaces in which the Firefleye system is used. The method required to do this will  
263 use position tracking by trajectory estimation. Position tracking of the light source  
264 will record the position, velocity, and acceleration of the projectile in (X,Y,Z)  
265 coordinates. By implementing  
266 position tracking, a quantifiable depth map will follow which will then lead to  
267 modeling of tunnels, caves, and subterranean features.  
268

## 269 **7 Conclusion**

270  
271 In this paper an investigation of the use of throwable lights for dark tunnel  
272 illumination was done. This was achieved by shooting spherical diffused light  
273 sources, to mimic ideal light source behavior, via the use of pressurized spud gun,  
274 and video capturing the flight trajectory of the light source.  
275

276 The data presented in previous sections show that, when firing at ideal  
277 pressure and angle, the projectile launcher achieved an average range of 60 feet  
278 with a standard deviation of 5 feet. It was discovered that the projectile light  
279 source became too dim within 20 min during testing due to issues with the drain  
280 rate of the batteries powering the onboard lights. Using the same LED lights and a  
281 bigger battery, a greater success was achieved lighting a larger tunnel. The lights  
282 in this case were mounted on a plane, and video capturing was done while human  
283 subject was running down the tunnel with the light source. The image fusion  
284 software surpassed established performance metrics.  
285

286 Testing in environments similar to lunar caves led to a conclusion with  
287 95% confidence that projectiles illuminate 90% of a cave approximately 6 ft high  
288 12 ft in width and 150 feet in depth compared to a long exposure image taken  
289 with a static cell phone flash light set behind the camera. Projectiles were  
290 launched a maximum distance of 200 ft on Earth, which translates to an estimated  
291 1200 ft on the moon. Testing conditions are similar enough to known properties  
292 of caves on the moon to show that the tests were valid for obtaining information  
293 about the composition of the walls of lunar caves as well as for improving  
294 navigation for a rover in these tunnels.  
295

## 296 **8 Acknowledgements**

297

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302

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304

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321

## 322 **10 Appendix**

323 Below are summary explanations of various concepts and ideas that were not thoroughly  
324 explained when mentioned in previous sections of this report.

### 325 **10.1 Software: Image Fusion and Depth Map**

#### 326 **10.1.1 Image fusion**

327 Image Fusion is a process of combining several images into one. This process is done in a way  
328 that the fused image would be more informative than each individual images used to create it.  
329 Several of the methods for image fusion are described below. Titles 2.1.2 through 2.1.5 are image  
330 fusion variants.

#### 331 **10.1.2 High Dynamic Range (HDR)**

332 High Dynamic Range (HDR) is a method of taking several low dynamic range images and  
333 creating one high dynamic range image. In photography dynamic range stands for the luminance  
334 range of the image. This method is used by many photographers in image post processing. Many  
335 image processing tools such as Photoshop, GIMP, Matlab, and more have this method integrated  
336 in them.

#### 337 **10.1.3 Mean/Average method**

338 The Mean/Average method is one of the simplest ways of performing image fusion. The process  
339 involves taking all input images and averaging each pixel over the inputs.

#### 340 **10.1.4 Weighted Average Method/Exposure Fusion**

341 This method is an improvement on the Mean/Average method and was proposed by T. Martin et.  
 342 al [1]. This method assumes that all the frames are perfectly aligned and computes the output  
 343 image by keeping only the “best” parts [2]. The fusion is done by first computing a weighted  
 344 average along each pixel in all frames using equation 1 then substituting that in equation 2.

$$345 \hat{W}_{i,j,k} = [\sum_{k'=1}^N W_{ij,k'}]^{-1} W_{ij,k'} \quad (1)$$

$$346 R_{i,j} = \sum_{k'=1}^N \hat{W}_{ij,k} I_{ij,k} \quad (2)$$

347 Where  $R_{i,j}$  is the fused image at pixel (i,j), W is the weight,  $\hat{W}$  is normalized weight, N is  
 348 number of frames, and  $I_k$  is the kth input frame[2].

### 349 **10.1.5 Image Fusion via PCA**

350 Principal Component Analysis (PCA) is a mathematical tool for performing a linear  
 351 transformation of data in N-dimensional space into a new set of coordinate systems called  
 352 principal components. Usually, the first principal component is computed in a way that it gives the  
 353 maximum variation along its axis. Skipping all the mathematical jargon, PCA boils down to an  
 354 Eigenvalue Eigenvector problem where the Eigenvector associated with the maximum Eigenvalue  
 355 gives the first principal component.

356 In terms of image fusion, one can take each input image as different dimensions (i.e. if one has N  
 357 frames then this becomes an N-dimensional problem where the pixels are data point in this high  
 358 dimensional space). Therefore, applying PCA on the frames would transform the data into new N  
 359 dimensional space where the first axis (first principal component) contains the majority of the  
 360 information (usually more than 90%). As a result, one can take this first component as the fused  
 361 image.

362 This is implemented in 4 easy steps

- 363 • Import all the images as column vectors and create one huge matrix, “im” (i.e. if we have  
 364 N frames that are Mpx by Kpx, then each frame would create an M\*K by 1 column  
 365 vector creating one huge matrix of M by N matrix).
- 366 • Compute the covariance matrix of matrix “im”. This computes the covariance between  
 367 each pixel over all the frames. This covariance matrix has dimensions N by N
- 368 • Now perform Singular Value Decomposition (SVD) on the covariance matrix. This returns  
 369 Eigenvectors (N by N matrix) and Eigenvalues (N by N matrix).
- 370 • Use the elements of the first Eigenvector (i.e. the first column for the Eigenvector Matrix)  
 371 as weights and do a linear combination of all the frames, which gives the fused image.

372 Sylvia et al. [2] published a paper in 2006 that used PCA based image fusion.

### 373 **10.1.6 Depth Map**

374 Depth map is an image that represents distances of different objects in a scene from a specified  
 375 view point. For the Fireflye project computing the depth map was possible due to the principle  
 376 related to Lambertian reflectance.

377 A Lambertian [3] object has a property that the apparent brightness of a surface to an observer is  
 378 the same regardless of the observer’s angle of view whereas the luminous intensity obeys the  
 379 Lambert’s cosine law. Lambert’s cosine law states that radiant intensity is proportional to the  
 380 cosine of the angle between the surface normal and observer’s line of sight [4].

## 381 **10.2 Mechanical**

382 No previous attempts at pneumatic launchers in space exist. Prior to this research, there has been

383 no reason to have portable launchers in a space environment. However, there are many different  
384 types of launching mechanisms that function on earth and can be applied to use in a lunar  
385 environment.

#### 386 **10.2.1 Flywheel**

387 A flywheel launcher uses two flywheels to launch a projectile, impulsively transferring their  
388 rotational energy to the projectile to propel it forward. This method is not practical for use in space  
389 because the entire system would necessarily be very heavy.

#### 390 **10.2.2 Coilgun**

391 A coil gun uses inductor coils to accelerate a ferrous projectile, which is reliable and variable in  
392 power but typically inefficient, which is not ideal in an environment where battery power is a  
393 measured resource.

#### 394 **10.2.3 Spring Launcher**

395 More realistically, a spring-loaded system requires little power. This kind of system is simple and  
396 reliable when designed properly and can function in space conditions very well.

#### 397 **10.2.4 Pneumatic Cannon**

398 Another viable option is a pneumatic launcher, which would use the expansion of compressed gas  
399 to propel a projectile down a barrel. This is a simple and reliable system if there is a source of  
400 compressible gas. Guaranteeing a source may be a problem in the absolute cold of a lunar  
401 environment, though compressed air has previously been used for unrelated (propulsion) purposes  
402 in space environments. While it is difficult to find discussion directly related to the effect of lunar  
403 conditions on maintaining a gaseous state pneumatic systems, the use of such systems in space  
404 exploration is definitely not unheard of.  
405

### 406 **10.3 Electrical**

407 The basic principle motivating launchable illumination is that the distance of a light  
408 source from an object largely determines the intensity of the light when it reaches that object. U.  
409 Wong suggested that the falloff with distance from any reasonable light was actually much more  
410 severe than that of an ideal point source, suggesting that mounting lights, however bright, directly  
411 on a rover is not a viable method for cave exploration as it limits how much information can  
412 reasonably be obtained about a particular path before attempting to traverse it [5]. Another way of  
413 avoiding this problem is the use of technologies such as LIDAR, which has been used for many  
414 other applications in robotics and space exploration [6]. While useful for generating depth maps  
415 of the lunar terrain, LiDAR cannot provide the information about cave wall composition and  
416 texture that traditional cameras provide. The shortcomings of these more traditional methods  
417 provide reasoning for light sources that can travel along a cave wall and produce clear images of  
418 their environment.

419

420